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(52) Thermoplastic elastomer-based low vulnerability ammunition gun propellants.

(57) LOVA gun propellants are formed from a thermoplastic elastomer and particulates of high-energy oxidizers, e.g., RDX and HMX.

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## THERMOPLASTIC ELASTOMER-BASED LOW VULNERABILITY AMMUNITION GUN PROPELLANTS

The present invention is directed to low vulnerability ammunition (LOVA) gun propellants in which the binder is a thermoplastic elastomer.

A continuing objective in the design of gun propellants is to provide a gun propellant which is energetic when deliberately ignited, but which exhibits high resistance to accidental ignition from heat, flame, impact, friction, and chemical action. Propellants possessing such resistance to accidental ignition are known as "low vulnerability ammunition" (LOVA) gun propellants.

Conventional LOVA gun propellants comprise an elastomeric binder, throughout which are dispersed particulates of high-energy material, particularly oxidizers. The elastomeric binder is generally a cured elastomer, formed, for example, by the urethane reaction of a multi-functional prepolymer with a multifunctional isocyanate. Examples of such LOVA gun propellants are described, for example, in U.S. Patents Nos. 4,263,070 and 4,456,493, the teachings of which are incorporated herein by reference. Generally, LOVA propellant grains are formed by extrusion at elevated temperatures whereat substantial curing takes place. Because the grains cure to some extent as they are being formed, control of extrusion conditions is difficult. If cured LOVA propellant is unused, it cannot be recycled, and burning the propellant is generally the only suitable disposal method.

Another type of LOVA propellant has a binder of cellulose acetate or a cellulose acetate derivative. An example of this type of propellant is described in U.S. Patent No. 4,570,540, the teachings of which are incorporated herein by reference. These types of LOVA propellants are solvent processed, a process which entails relatively long processing times and a large number of steps. Also, the use of solvent creates environmental problems.

The present invention is directed to LOVA propellants which use thermoplastic elastomers as binders. Thermoplastic elastomers have been previously used in propellants for rocket motors or the like, for example, as described in U.S. Patent No. 4,361,526 and U.S. Patent Application S.N. 06/925,660 filed October 29, 1986, the teachings of each being incorporated herein by reference. Gun propellants, however, are considered to be a different art than rocket motor propellants. Rocket motor propellants typically contain a particulate metal fuel, e.g., particulate aluminum. Gun propellants, on the other hand, should be substantially free of any metal, and for that matter, should be generally free of any material which leaves a solid residue in the barrel of the gun upon burning. Gun propellants should also be substantially free of chlorine, which degrades the gun barrel.

Furthermore, rocket motor grains are typically formed in a different manner. Gun propellant grains typically take their shape from the extrusion process and must be sufficiently solid when leaving the extruder to retain their extruded shape. Material for rocket motor propellants may be extruded, but generally large rocket motors assume their shape from a mold, e.g., the rocket motor case; thus, after leaving an extruder or mixer, a propellant composition for a rocket motor should be free-flowing or at least moldable so as to be able to assume the shape of the large mold.

SUMMARY OF THE INVENTION

In accordance with the present invention, LOVA gun propellants comprise between about 60 and about 85 wt. percent of high-energy oxidizer particulates and between about 15 and about 40 wt. percent of a binder system which is a plasticized or unplasticized block copolymer having at least one crystalline block and at least one amorphous block, giving the block copolymer thermoplastic elastomeric characteristics.

DETAILED DESCRIPTION OF CERTAIN PREFERRED EMBODIMENTS

LOVA gun propellants comprise between about 60 and about 85 wt. percent of a high-energy oxidizer particulates and between about 15 and about 40 wt. percent of an elastomeric, thermoplastic binder system. The thermoplastic elastomer of the binder system has at least one block which is amorphous at room temperature, e.g., in the range of about 20 °C to about 25 °C and at least one block which is crystalline at room temperature. It is generally necessary that in the block copolymer molecule, there be at least a pair of crystalline blocks flanking an amorphous block, whereby a thermoplastic network may be formed. The crystalline hard blocks preferably melt in a temperature range of between about 70 °C and about 105 °C. This temperature range allows processing at temperatures which do not decompose the nitramine fillers. At

the same time, in this temperature range, the binder retains good mechanical properties at about 63°C, considered to be the upper use temperature of LOVA gun propellants. The binder system may contain up to about 80 wt. percent of an energetic or non-energetic plasticizer, the plasticizer comprising up to about 35 wt. percent of the LOVA gun propellant composition as a whole.

5 The two most common oxidizer particulates are tetramethylenetrinitramine (HMX) and trimethylenetrinitramine (RDX). Mixtures of these oxidizers may be used.

Various configurations of thermoplastic elastomers are suitable, including (AB)<sub>n</sub> polymers, ABA polymers, and A<sub>n</sub>B star polymers, wherein the A blocks are crystalline and B blocks are amorphous at room temperature. In each of these structures, at least two A blocks flank at least one B block, allowing the 10 crystalline A blocks to define a cross-linked structure at lower temperatures, while the amorphous B blocks give the polymer its elastomeric properties.

A wide variety of thermoplastic elastomers may be used in accordance with the present invention, including polyoxetanes, mixed polyesters, polyester-polyethers, and polyamide-polyethers. ABA polymers based upon polyoxetanes and poly(oxetane/tetrahydrofuran) copolymers are described in the above-referenced U.S. Patent Application S.N. 06/925,660. (AB)<sub>n</sub> polymers based upon polyoxetanes and poly(oxetane-tetrahydrofuran) copolymers are described in U.S. Patent Application No. 07/174,665, filed March 29, 1988, the teachings of which are incorporated herein by reference. Another specific thermoplastic elastomers is polyethylene succinate/poly diethyleneglycol adipate (PES/PEDGA) block polymers.

Currently preferred thermoplastic polymers are (AB)<sub>n</sub> type polyesters having short chain crystalline ester units and long chain amorphous ester units. Examples of such polymers are:

	Polyester Number	Short Chain Ester Units	Long Chain Ester Units
25	1	4GI	PTMEG
	2	4GI	PEG
	3	6GT/4GT	PTMEG
	4	6GT/6GI	PTMEG
	5	6GT/4GT	PEG
	6	4GT/4GI	PEG
30	4GI	1,4-butylene isophthalate	
	4GT	1,4-butylene terephthalate	
	6GI	1,6-butylene terephthalate	
	6GT	1,6-butylene terephthalate	
35	PTMEG	polytetramethylene ether glycol	
	PEG	polyethylene ether glycol	

40 The plasticizer, if used, may be non-energetic, e.g., dioctyl phthalate (DOP), dioctyl adipate (DOA), Santicizer 8 polyester by Monsanto, butanetriol trinitrate (BTTN), trimethylolethane trinitrate (TMETN), polyglycidal nitrate, or nitroglycerine (NG). Generally, if an energetic plasticizer is used, it is used at a low level in order to maintain the low vulnerability properties of the propellant. Other suitable plasticizers include, but are not limited to dibutoxyethyl phthalate (DBEP), dibutoxyethyl adipate (DBEA), chlorinated 45 paraffin, methyl abietate, methyl dihydro-abietate, n-ethyl-o and p-toluene sulfonamide, polypropylene glycol sebacate, dipropylene glycol dibenzoate, di(2-ethyl-hexyl) phthalate, 2-ethyl-hexyl-diphenyl phosphate, tri(2-ethyl-hexyl) phosphate, di(2-ethyl-hexyl)sebacate, Santicizer 409 polyester by Monsanto, tetra-ethylene glycol-di(2-ethyl hexoate), dibutoxyethoxyethyl adipate (DBEEA), oleamide, dibutoxyethyl azelate (DBEZ), dioctyl azelate (DOZ), dibutoxyethoxyethyl glutarate (DBEEG), dibutoxyethyl glutarate (DBEG), polyethylene glycol 400 dilaurate, polyethylene glycol 400 dioleate, dibutoxyethoxyethyl sebacate, dibutoxyethyl sebacate, and trioctyl trimellitate (TOTM).

The thermoplastic elastomer must be selected so that the filled propellant has a strain (elongation) of at least 1 percent, preferably at least about 3 percent, and preferably less than 10. The modulus must be high enough so that the propellant grain maintains its shape during firing, i.e., so that it does not compress into a blob, and sufficiently low so as not to be brittle. A relatively broad range of moduli are acceptable, i.e., a range of between about 5,000 and about 50,000, preferably below about 35,000.

Propellant compositions are generally required to operate over a wide temperature range and gun propellant grains should be stable at least to a temperature of 165°F (74°C). In order for the gun propellants to be used in low temperature environments, it is preferred that the thermoplastic elastomers

incorporate soft blocks which retain their amorphous characteristics at low temperatures, i.e., down to -20 °C and, preferably, even down to -40 °C. Gun propellant grains are generally intended to operate in high pressure ranges, i.e., 30,000 psi or above.

In addition to the binder system and the oxidizer particulates, the LOVA gun propellant composition 5 may contain minor amounts of other materials, such as processing aids, lubricants, colorants, etc.

An important difference between rocket motor propellants and gun propellants is that gun propellants are fired through a barrel which is used multiple times, requiring that the gun propellants be substantially free of materials which would either corrode the barrel or leave deposits in the barrel. Gun propellants are substantially free of metallic particulates and other materials which leave a solid residue. Generally, metal-containing compounds are avoided as these tend to leave deposits; however, metal in compound form may 10 comprise up to about 0.5 wt. percent of the total weight of the propellant composition. For example, potassium sulfate may be incorporated as a flame suppressant. To avoid gun barrel corrosion, corrosive materials or materials which become corrosive upon firing are avoided. Gun propellants should be substantially free of chlorine.

The propellants are processed by blending the ingredients at a temperature of between about 100 °C 15 and 125 °C in a mixer, such as a horizontal sigma blade mixer, planetary vertical mixer or twin screw mixer. The mix is then extruded and cut into a predetermined shape. Extrusion temperatures typically range from about 70 °C to 130 °C. A typical shape for a gun propellant is a cylinder having a plurality of axially-directed perforations. In one typical embodiment, the propellant is cylindrical having a perforation running along the 20 cylindrical axis and six additional perforations arranged along a circle halfway between the central perforation and the outside cylindrical wall.

One general feature of thermoplastic elastomers which makes them particularly suitable for LOVA gun propellant applications is their endothermic melting characteristics. The fact that they absorb thermal energy as they begin to melt makes the LOVA gun propellants more capable of withstanding high 25 temperatures.

The invention will now be described in greater detail by way of specific examples.

#### EXAMPLE 1

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Table 1 below summarizes various properties of LOVA gun propellants prepared using different thermoplastic elastomeric binder systems, including mixing conditions, extrusion conditions, mechanical and physical properties and burn rates. In each case, the composition is 78% RDX, 22% binder system. The 35 third composition from the left has a binder system which includes 20% by weight of a non-energetic plasticizer, diethyl phthalate (DOP). The fourth polymer is of the type reported in above-identified U.S. Patent Application S.N. 06/925,660 as being an ABA block polymer wherein poly(3,3-bis(azidomethyl)-oxetane) (BAMO) forms the crystalline A blocks and wherein the B block is a copolymer of poly(3,3-bis(azidomethyloxetane)/3-azidomethyl-3-methyloxetane) (BAMO/AMMO).

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TABLE I

Polymer	PES PDEGA	Polyester #1 (4GI/PTMEG) Santicizer 8	Polyester #1 (4GI/PTMEG) DOP(4:1)	B-B/A-B
Rheocord 40 Test (78% RDX)	LT035	LT033	LT051	LT049
Peak Torque, m-g	590	416	1255	971
Peak Temperature, °C	116	114	128	119
Extrusion (EX87)	0707-2	0629	0930-2	0921-2
600 psi Barrel T, °C	89 ° (750psi)	95	112	85
Die T, °C	80	85	99	78
DSC (10 °C/min, N <sub>2</sub> )				
Tg, °C	-44	-54	-35	-41
Tm, °C	+ 79	+ 93	+ 120	+ 93
63 °C Slump				
Compressibility, %	2.2	19	1.9	2.2
60 Min Creep, %	1.6	17	0.3	1.2
DMA (5 °C/Min)				
Tg, °C	-33	-39	-64	-24
E @ -40 °C, MPa	568	508	343	763
0 °	224	89	201	315
+ 20 °	151	55	162	195
+ 40 °	55	9	99	118
Tensile @ 25 °C (0.1 in/min)				
Modulus, psi	14,000	6000	25,300	21,000
Stress, psi	234	59	460	235
Strain, %	2.2	1.1	2.0	1.3
Burn Rate @ 11,000 psi, in/sec	0.85	1.10	0.76	1.88
26,000 psi, in/sec	2.89	4.09	2.09	4.82

EXAMPLE 2

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Table 2 below summarizes properties of LOVA gun propellants prepared from various (AB)<sub>n</sub> block polymers having oxetane and tetrahydrofuran (THF) mer units. In each case, BEMO comprises the crystalline blocks. The soft blocks are oxetane polymers, oxetane copolymers, and oxetane/THF copolymers. NMMO is an abbreviation for poly(3-nitratomethyl-3-methyloxetane). BMMO is an abbreviation for poly(3,3-bis(methoxymethyl)oxetane). The (AB)<sub>n</sub> polymers are described in above-referenced U.S. Patent Application S.N. 07/174,665.

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TABLE II

	Polymer	TPE-1	ETPE-2	ETPE-4	ETPE-5
5	Soft block Lot No. RBW <u>Rheocord Test (78%)RDX</u>	BMMO/THF III-56 LT026	BAMO/AMMO IV-24 LT048	NMMO IV-12 LT039	BAMO/NMMO IV-10 LT037
10	Peak Torque, m-g Peak Temperature, °C Extrusion (EX87) 600 psi. Barrel T, °C Die T, °C	1358 118° 0521 86° 79°	1089 120° 0921-1 86° 79°	780 120° 0825-1 94° 86°	1044 121° 0810 90° 84°
15	DSC (10 °C/min, N <sub>2</sub> )				
20	Tg, °C Tm, °C 63 °C Slump	-47° +69°	-36° +79°	-25° +75°	-28° +76°
25	Compressibility, % 60 Min. Creep, %	2.4 1.0	2.6 0.5	1.6 0.6	1.3 0.5
30	DMA (5 °C/Min)				
35	Tg, °C E @ -40 °C, MPa 0° +20° +40° Tensiles @ 25 °C (0.1 in/min)	-30° 553 265 159 64	-21° 600 342 214 126	-11° 627 440 185 100	-13° 613 447 194 97
40	Modulus, psi Stress, psi Strain, psi <u>Burn Rate @ 11,000 psi, in/sec</u> 26,000 psi, in/sec	29,000 261 2.3 0.83 2.33	31,000 375 1.6 1.10 2.96	29,000 408 1.9 1.06 3.02	24,000 461 2.0 1.12 3.12
45	Drop Wt., Mech. Props.				
50	Strain rate, sec <sup>-1</sup> Modulus, GPa Failure Stress, MPa Strain, %	312 1.92 40.7 4.26		274 2.28 51.5 3.32	282 3.12 60.7 3.00

Thermoplastic elastomers of the (AB)<sub>n</sub> type suitable for forming gun propellants in accordance with the present invention may be made from joining hard blocks and soft blocks from the following lists in the manner taught in the above-referenced U.S. Patent Application No. 07/174,665:

Soft Blocks

- 50 poly ethylene glycol (PEG)
- polycaprolactone (PCP)
- polytetrahydrofuran (PolyTHF)
- polypropylene glycol (PPG)
- amorphous polyoxetanes
- 55 poly(ethylene oxide-tetrahydrofuran)
- poly(diethylene glycol adipate)
- polyglycidyl nitrate
- polyglycidyl azide (GAP)

Hard Blocks

- polyallyl acrylate
- polyisobutyl acrylate
- 5 poly 1,4-cyclohexylenedimethylene formal, trans
- poly 1,2-cyclopropanedimethylene isophthalate
- poly decamethylene adipate
- poly decamethylene azelaate
- poly decamethylene oxalate
- 10 poly decamethylene sebacate
- polyethylene sebacate
- polyethylene succinate
- poly hexamethylene sebacate
- poly 10-hydroxydecanoic acid
- 15 poly tert-butyl-isotactic
- poly nonamethylene terephthalate
- poly octadecamethylene terephthalate
- poly 3,3-bisethoxymethyl (BEMO)
- poly pentamethylene terephthalate
- 20 poly B-propiolactone
- poly tetramethylene p-phenylenediacetate
- poly trimethylene oxalate
- polyethyl vinyl ether
- polypropyl vinyl ether
- 25 poly p-xylylene adipate
- poly p-xylylene sebacate.

**Claims**

30 1. A low vulnerability ammunition gun propellant composition comprising from about 60 to 85 wt. percent of particulates of a high-energy oxidizer and between about 15 wt. percent and about 40 wt. percent of a thermoplastic, elastomeric binder system, said binder system being substantially free of metallic particulates and materials which leave a solid residue, said binder system comprising a thermoplastic elastomer and from 0 to about 80 wt. percent of a plasticizer.

35 2. A propellant composition according to Claim 1 wherein a non-energetic plasticizer is used.

3. A propellant composition according to Claim 2 wherein said non-energetic plasticizer is diethyl phthalate.

40 4. A propellant composition according to Claim 1 wherein an energetic plasticizer is used.

5. A propellant composition according to Claim 4 wherein said plasticizer is selected from the group consisting of butanetriol trinitrate, trimethylolethane trinitrate and nitroglycerine.

6. A propellant composition according to Claim 1 wherein the oxidizer from which said oxidizer particulates are formed is selected from the group consisting of tetramethylenetrinitramine, trimethylenetrinitramine, and mixtures thereof.

45 7. A propellant composition according to Claim 1 wherein said thermoplastic elastomer is a block polymer having polyethylene succinate blocks and polydiethyleneglycoladipate blocks.

8. A propellant composition according to Claim 1 wherein said thermoplastic elastomers are formed from blocks of oxetane polymers, oxetane copolymers and oxetane/tetrahydrofuran copolymers.

9. A propellant according to Claim 8 wherein crystalline blocks of said thermoplastic elastomer are poly-(3,3-bis(ethoxymethyl) oxetane).

50 10. A propellant according to Claim 8 wherein amorphous blocks of said thermoplastic elastomers are selected from the group consisting of poly((3-nitratomethyl-3-methyl)oxetane), poly(3,3-bis(azidomethyl)-oxetane/3-azidomethyl-3-methyl) oxetane), poly(3,3-bis(azidomethyl)oxetane/3-nitratomethyl-3-methyl)-oxetane), and poly(3,3-bis(methoxymethyl)oxetane/tetrahydrofuran).

55 11. A propellant composition according to Claim 1 wherein said thermoplastic elastomer is a polyester having hard blocks formed from the group consisting of 1,4-butylene isophthalate, 1,4-butylene terephthalate, 1,6-butylene isophthalate, 1,6-butylene terephthalate and mixtures thereof and soft blocks selected from the group consisting of polytetramethylene ether glycol, polyethylene ether glycol and mixture thereof.

12. A propellant in accordance with Claim 10 wherein said propellant is substantially free of chlorine.
13. A propellant in accordance with Claim 1 wherein said crystalline blocks of said thermoplastic elastomer melt in a temperature range of between about 70 °C and about 105 °C.

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## EUROPEAN SEARCH REPORT

Application Number

EP 89 30 7718

DOCUMENTS CONSIDERED TO BE RELEVANT									
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	CLASSIFICATION OF THE APPLICATION (Int. Cl.5)						
D, P X	US-A-4 806 613 (R.B. WARDLE) * Column 1, lines 42-60; column 6, line 36 - column 7, line 20; claims *	1, 4-6, 8 -10, 12, 13	C 06 B 45/10 C 06 B 25/34						
P, X	US-A-4 764 316 (E.D. BROWN et al.) * Column 5, lines 11-33; column 9, lines 55-63 *	1-3, 6							
X	US-A-3 265 543 (ALBERT SMITH CARTER) * Claims *	1, 4							
D, A	US-A-4 361 526 (THE UNITED STATES OF AMERICA) * Column 1, line 12 - column 2, line 56 *	2, 3, 6							
X	DE-B-1 110 065 (E.I. DU PONT DE NEMOURS AND CO.) * Column 3, lines 6-21 *	1							
A	GB-A-1 362 506 (ROCKWELL INTERNATIONAL CORP.) * Page 2, lines 32-38 *	12	TECHNICAL FIELDS SEARCHED (Int. Cl.5)						
A	"ENCYCLOPEDIA OF POLYMER SCIENCE AND ENGINEERING", vol. 5, Dielectric Heating to Embedding, 1986, John Wiley & Sons, New York, US; * Pages 416-419 *	1, 7, 8, 11	C 06 B						
A	"ENCYCLOPEDIA OF POLYMER SCIENCE AND ENGINEERING", vol. 12, Polyesters to Polypeptide Synthesis, 1988, John Wiley & Sons, New York, US; * Pages 49-52, 75-77 *	7, 11							
		-/-							
<p>The present search report has been drawn up for all claims</p> <table border="1"> <tr> <td>Place of search</td> <td>Date of completion of the search</td> <td>Examiner</td> </tr> <tr> <td>THE HAGUE</td> <td>13-01-1992</td> <td>SCHUT, R.J.</td> </tr> </table>				Place of search	Date of completion of the search	Examiner	THE HAGUE	13-01-1992	SCHUT, R.J.
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CATEGORY OF CITED DOCUMENTS		T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document							
X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document									



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### CLAIMS INCURRING FEES

The present European patent application comprised at the time of filing more than ten claims.

- All claims fees have been paid within the prescribed time limit. The present European search report has been drawn up for all claims.
- Only part of the claims fees have been paid within the prescribed time limit. The present European search report has been drawn up for the first ten claims and for those claims for which claims fees have been paid.  
namely claims:
- No claims fees have been paid within the prescribed time limit. The present European search report has been drawn up for the first ten claims.

### LACK OF UNITY OF INVENTION

The Search Division considers that the present European patent application does not comply with the requirement of unity of invention and relates to several inventions or groups of inventions.

namely:

see sheet -B-

- All further search fees have been paid within the fixed time limit. The present European search report has been drawn up for all claims.
- Only part of the further search fees have been paid within the fixed time limit. The present European search report has been drawn up for those parts of the European patent application which relate to the inventions in respect of which search fees have been paid.  
namely claims:
- None of the further search fees has been paid within the fixed time limit. The present European search report has been drawn up for those parts of the European patent application which relate to the invention first mentioned in the claims.

namely claims: 1-7, 12, 13



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## EUROPEAN SEARCH REPORT

Page 2

Application Number

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DOCUMENTS CONSIDERED TO BE RELEVANT			CLASSIFICATION OF THE APPLICATION (Int. Cl.5)						
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim							
A	"ENCYCLOPEDIA OF POLYMER SCIENCE AND ENGINEERING", vol. 2, Anionic Polymerization to Cationic Polymerization, 1985, John Wiley & Sons, New York, US; * Pages 398-400 *	11							
			TECHNICAL FIELDS SEARCHED (Int. Cl.5)						
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The Search Division considers that the present European patent application does not comply with the requirement of unity of invention and relates to several inventions or groups of inventions,

namely:

1. Claims 1-7,12,13 : Thermoplastic elastomeric binder system I.
2. Claims 1-6,8-10,12-13 : Thermoplastic elastomeric binder system II.
3. Claims 1-6,11-13 : Thermoplastic elastomeric binder system III.

Three chemically quite different binder systems. See also corresponding US patents US-A-4 976 794, US-A-4 919 737.